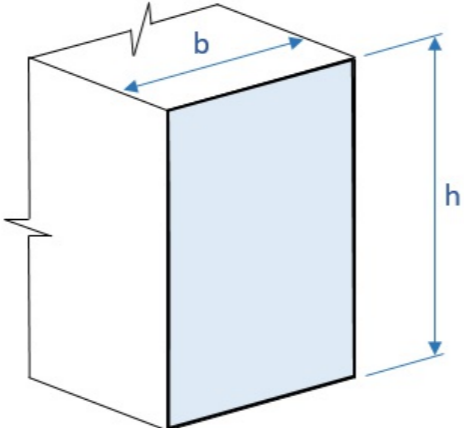


| REFERENCES | CALCULATIONS | RESULTS |
|-------------------------|--|--------------------------|
| <p>Code: ACI 318-14</p> | <p>MEMBER #1 (SECTION POSITION 120.0 INCHES) COLUMN DESIGN REPORT</p> <p>Project details</p> <p>Project Name: Not Provided Project ID: Not Provided Company: Not Provided Designer: Not Provided Client: Not Provided Project Notes: Not Provided Project Units: Imperial</p> <p>General member design information</p> <p>Dimensions:</p>  <p>Height $h = 16$ in Width $b = 16$ in Member length = 240 in</p> <p>Material properties: Concrete strength $f_c = 5000$ psi Steel strength of longitudinal rebar $f_y = 60000$ psi Steel strength of shear rebar $f_{yt} = 50000$ psi Permissible crack width $c_w = 0.012$ in</p> <p>Load Combinations (Ultimate Limit State)</p> <p>For axial force in section: LC1: USER = 100 Kip</p> <p>For bending moment in section (major axis): LC1: USER = 10 Kip-ft</p> <p>For bending moment in section (minor axis): LC1: USER = 10 Kip-ft</p> <p>Load Combinations (Serviceability Limit State)</p> <p>For axial force in section: LC1: USER = 0 Kip</p> <p>For bending moment in section (major axis): LC1: USER = 0 Kip-ft</p> <p>For bending moment in section (minor axis): LC1: USER = 0 Kip-ft</p> | |
| | <p>Detailing of Members</p> <p>DETAILING RULES FOR COLUMN (LONGITUDINAL REINFORCEMENT)</p> <p>Section input data: Section concrete area $A_g = 256.00$ in² Longitudinal rebar area $A_{st} = 8.00$ in²</p> <p>Check the steel ratio for the longitudinal steel:</p> $\rho = \frac{A_{st}}{A_g} = \frac{8.00}{256.00} = 0.03125$ <p style="text-align: center;">$0.01 < 0.03125 < 0.08$</p> | <p>STATUS OK!</p> |

Column check

MAXIMUM AXIAL COMPRESSION

Section input data:

Section concrete area $A_g = 256.00 \text{ in}^2$

Longitudinal rebar area $A_{st} = 8.00 \text{ in}^2$

Calculate the axial load capacity for concentric loading

$$\begin{aligned} P_u &= \phi \cdot P_n = \phi \cdot K \cdot \left[0.85 \cdot f'_c \cdot (A_g - A_{st}) + f_y \cdot A_{st} \right] = \\ &= 0.65 \cdot 0.8 \cdot \left[0.85 \cdot f'_c \cdot (A_g - A_{st}) + f_y \cdot A_{st} \right] = \\ &= 0.65 \cdot 0.8 \cdot \left[0.85 \cdot 5000 \cdot (256.00 - 8.00) + 60000 \cdot 8.00 \right] \cdot 0.001 = 797.68 \text{ Kip} \end{aligned}$$

MAXIMUM AXIAL TENSION

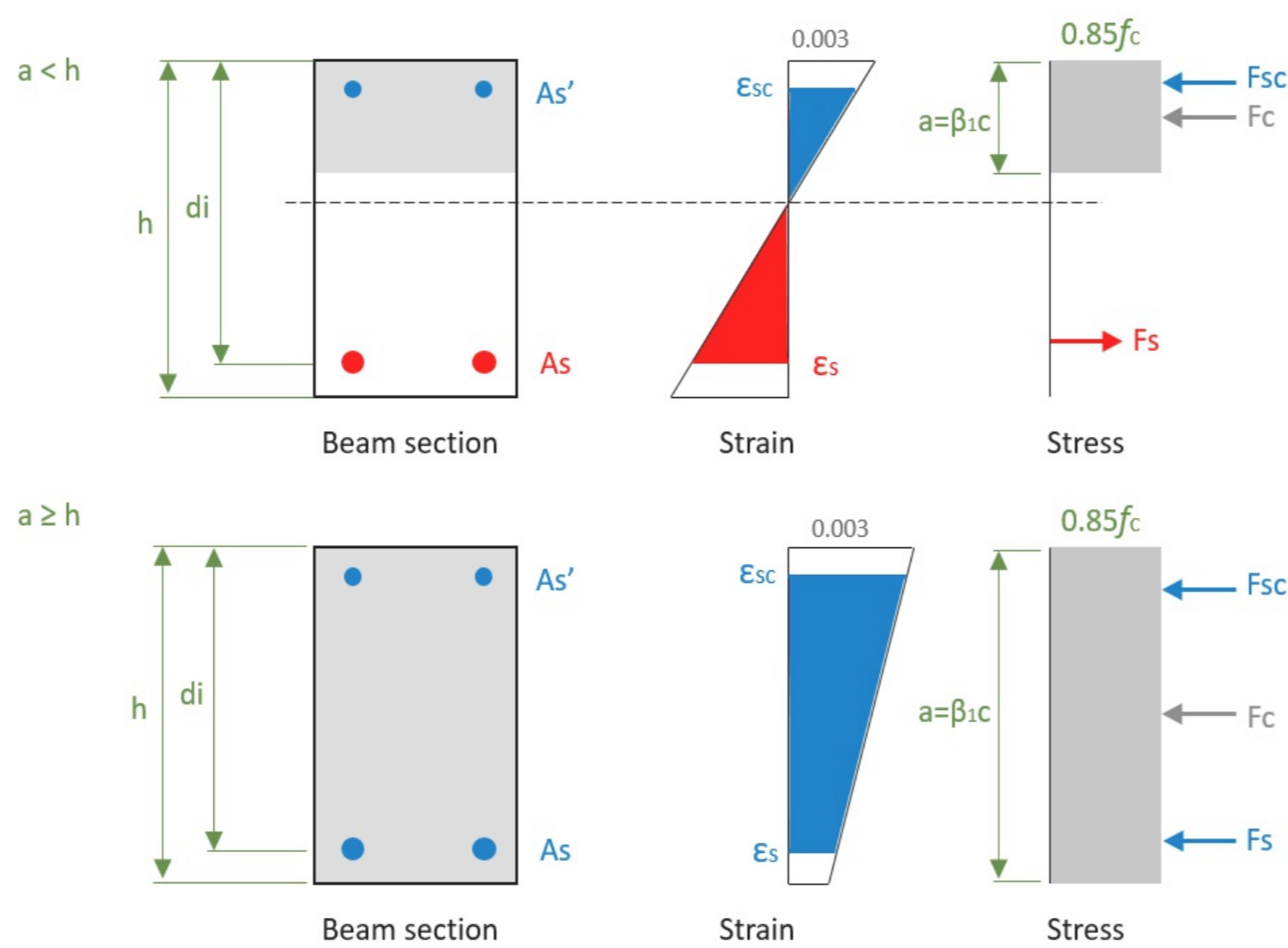
Section input data:

Longitudinal rebar area $A_{st} = 8.00 \text{ in}^2$

Calculate the axial load capacity for concentric loading

$$T_n = -0.9 \cdot f_{yd} \cdot A_{st} = -0.9 \cdot 60000 \cdot 8.00 \cdot 0.001 = -432.00 \text{ Kip}$$

M-N INTERACTION



Calculation is based on iterative process:

- Calculate plastic centroid location \bar{x}_p
- Assume c in tension control zone and compression control zone
- Calculate strain e_s and e_{sc} when $a < h$:

$$e_{sc} = e_c \cdot ((c - d)/c)$$

$$e_s = e_c \cdot ((d - c)/c)$$

and $a > h$:

$$e_{sc} = 0.002 \cdot (7 \cdot (c - d) / (7 \cdot c - 3 \cdot h))$$

$$e_s = 0.002 \cdot (7 \cdot (c - d) / (7 \cdot c - 3 \cdot h))$$

- Calculate reinforcement stresses $f_s = \{e_s E_s (e_s \leq e_y), e_y (e_s > e_y)\}$

- Calculate equilibrium forces:

$$N = F_{cc} + F_{sc} + F_s$$

$$a < h: N = f'_c \cdot b \cdot \beta_1 \cdot x + \sum f_{sci} \cdot A_{si} + \sum f_{si} \cdot A_{si}$$

$$a \geq h: N = f'_c \cdot b \cdot h + \sum f_{sci} \cdot A_{si} + \sum f_{si} \cdot A_{si}$$

$$a < h: M = F_{cc} \cdot (\bar{x}_p - \beta_1 x / 2) + \sum f_{sci} \cdot (\bar{x}_p - d_i) - \sum f_{si} \cdot (d_i - \bar{x}_p)$$

$$a \geq h: M = F_{cc} \cdot (\bar{x}_p - h / 2) + \sum f_{sci} \cdot (\bar{x}_p - d_i) - \sum f_{si} \cdot (d_i - \bar{x}_p)$$

1. Axial + positive flexure about major axis

Section input data:

Section height h based on major axis: 16 in

Section width b based on major axis: 16 in

Rectangular compression block factors:

$$\begin{aligned} 4000 \text{ psi} \leq f_c \leq 8000 \text{ psi} \rightarrow \beta_1 &= 0.85 - \left(\frac{f_c - 4000}{1000} \right) \cdot 0.05 = 0.85 - \left(\frac{5000 - 4000}{1000} \right) \\ &\cdot 0.05 = 0.80 \end{aligned}$$

Section Rebar

| Depth di (in) | bar diameter (in) | bar area Asi (in ²) |
|---------------|-------------------|---------------------------------|
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |

Calculation of plastic centroid location

$$\bar{x}_p = \frac{f'_c \cdot A_g \cdot (h - \text{neutral axis}) + \sum F_{sci} \cdot a_{sci} + \sum F_{si} \cdot a_{si}}{f'_c \cdot A_g + f_{sy} \cdot A_s + f_{sy} \cdot \dot{A}_s} =$$

$$= \frac{5000.00 \cdot 256.00 \cdot (16 - 8) + 3240000.00 + 600000.00}{5000.00 \cdot 256.00 + 60000.00 \cdot 4 + 60000.00 \cdot 4.00} = 8.00 \text{ in}$$

M-N interaction values

| Iter. | c (in) | a (in) | φ | Pn (Kip) | Mn (Kip-ft) | φPn (Kip) | φMn (Kip-ft) | e (in) |
|-----------------|--------|--------|------|----------|-------------|-----------|--------------|--------|
| Tension Control | | | | | | | | |
| 1 | - | - | 0.9 | -480.0 | 0 | -432.0 | 0 | 0 |
| 2 | 0.32 | 0.26 | 0.90 | -462.6 | 11.4 | -416.4 | 10.3 | -0.3 |
| 3 | 0.64 | 0.51 | 0.90 | -445.2 | 22.4 | -400.7 | 20.2 | -0.6 |
| 4 | 0.96 | 0.77 | 0.90 | -427.8 | 33.1 | -385.1 | 29.8 | -0.9 |
| 5 | 1.28 | 1.02 | 0.90 | -410.5 | 43.4 | -369.4 | 39.1 | -1.3 |
| 6 | 1.60 | 1.28 | 0.90 | -349.5 | 73.3 | -314.6 | 66.0 | -2.5 |
| 7 | 1.92 | 1.53 | 0.90 | -241.4 | 124.4 | -217.3 | 112.0 | -6.2 |
| 8 | 2.24 | 1.79 | 0.90 | -159.2 | 163.3 | -143.3 | 147.0 | -12.3 |
| 9 | 2.56 | 2.05 | 0.90 | -93.2 | 194.4 | -83.9 | 175.0 | -25.0 |
| 10 | 2.88 | 2.30 | 0.90 | -38.0 | 220.2 | -34.2 | 198.2 | -69.5 |
| 11 | 3.20 | 2.56 | 0.90 | 9.6 | 242.1 | 8.7 | 217.9 | 301.5 |
| 12 | 3.52 | 2.81 | 0.90 | 51.8 | 261.2 | 46.6 | 235.0 | 60.5 |
| 13 | 3.84 | 3.07 | 0.90 | 89.8 | 277.9 | 80.8 | 250.1 | 37.2 |
| 14 | 4.15 | 3.32 | 0.90 | 124.6 | 292.9 | 112.2 | 263.6 | 28.2 |
| 15 | 4.47 | 3.58 | 0.90 | 157.0 | 306.3 | 141.3 | 275.7 | 23.4 |
| 16 | 4.79 | 3.84 | 0.90 | 187.3 | 318.5 | 168.6 | 286.7 | 20.4 |
| 17 | 5.11 | 4.09 | 0.89 | 216.0 | 329.6 | 193.0 | 294.4 | 18.3 |
| 18 | 5.43 | 4.35 | 0.85 | 243.4 | 339.6 | 207.8 | 289.9 | 16.7 |
| 19 | 5.75 | 4.60 | 0.82 | 269.7 | 348.8 | 220.7 | 285.4 | 15.5 |
| 20 | 6.07 | 4.86 | 0.79 | 295.1 | 357.2 | 232.1 | 280.9 | 14.5 |
| 21 | 6.39 | 5.11 | 0.76 | 319.6 | 364.8 | 242.3 | 276.6 | 13.7 |
| 22 | 6.71 | 5.37 | 0.73 | 343.5 | 371.8 | 251.5 | 272.3 | 13.0 |
| 23 | 7.03 | 5.62 | 0.71 | 366.8 | 378.1 | 260.0 | 268.1 | 12.4 |
| 24 | 7.35 | 5.88 | 0.69 | 389.5 | 383.9 | 267.8 | 263.9 | 11.8 |

| | | | | | | | | |
|---------------------|-------|-------|------|--------|-------|-------|-------|------|
| 25 | 7.67 | 6.14 | 0.67 | 411.8 | 389.0 | 275.1 | 259.9 | 11.3 |
| 26 | 7.99 | 6.39 | 0.65 | 433.8 | 393.6 | 281.9 | 255.8 | 10.9 |
| Compression Control | | | | | | | | |
| 27 | 8.31 | 6.65 | 0.65 | 474.7 | 385.8 | 308.6 | 250.7 | 9.8 |
| 28 | 8.63 | 6.90 | 0.65 | 513.2 | 377.9 | 333.6 | 245.6 | 8.8 |
| 29 | 8.95 | 7.16 | 0.65 | 550.1 | 370.4 | 357.6 | 240.8 | 8.1 |
| 30 | 9.27 | 7.42 | 0.65 | 585.6 | 363.1 | 380.7 | 236.0 | 7.4 |
| 31 | 9.59 | 7.67 | 0.65 | 620.0 | 356.0 | 403.0 | 231.4 | 6.9 |
| 32 | 9.91 | 7.93 | 0.65 | 653.3 | 349.1 | 424.6 | 226.9 | 6.4 |
| 33 | 10.23 | 8.19 | 0.65 | 685.5 | 342.2 | 445.6 | 222.4 | 6.0 |
| 34 | 10.55 | 8.44 | 0.65 | 716.9 | 335.3 | 466.0 | 218.0 | 5.6 |
| 35 | 10.87 | 8.70 | 0.65 | 747.5 | 328.5 | 485.8 | 213.5 | 5.3 |
| 36 | 11.19 | 8.96 | 0.65 | 777.3 | 321.6 | 505.2 | 209.0 | 5.0 |
| 37 | 11.51 | 9.21 | 0.65 | 806.4 | 314.7 | 524.1 | 204.5 | 4.7 |
| 38 | 11.83 | 9.47 | 0.65 | 834.8 | 307.7 | 542.6 | 200.0 | 4.4 |
| 39 | 12.16 | 9.72 | 0.65 | 862.7 | 300.6 | 560.8 | 195.4 | 4.2 |
| 40 | 12.48 | 9.98 | 0.65 | 890.1 | 293.3 | 578.6 | 190.7 | 4.0 |
| 41 | 12.80 | 10.24 | 0.65 | 916.9 | 285.9 | 596.0 | 185.9 | 3.7 |
| 42 | 13.12 | 10.49 | 0.65 | 943.3 | 278.4 | 613.2 | 181.0 | 3.5 |
| 43 | 13.44 | 10.75 | 0.65 | 969.3 | 270.7 | 630.1 | 175.9 | 3.4 |
| 44 | 13.76 | 11.01 | 0.65 | 994.9 | 262.8 | 646.7 | 170.8 | 3.2 |
| 45 | 14.08 | 11.26 | 0.65 | 1020.1 | 254.6 | 663.1 | 165.5 | 3.0 |
| 46 | 14.40 | 11.52 | 0.65 | 1045.0 | 246.3 | 679.2 | 160.1 | 2.8 |
| 47 | 14.72 | 11.77 | 0.65 | 1069.5 | 237.8 | 695.2 | 154.5 | 2.7 |
| 48 | 15.04 | 12.03 | 0.65 | 1093.7 | 229.0 | 710.9 | 148.8 | 2.5 |
| 49 | 15.36 | 12.29 | 0.65 | 1117.7 | 219.9 | 726.5 | 143.0 | 2.4 |
| 50 | 15.68 | 12.54 | 0.65 | 1141.3 | 210.7 | 741.9 | 136.9 | 2.2 |
| 51 | 16.00 | 12.80 | 0.65 | 1164.8 | 201.1 | 757.1 | 130.7 | 2.1 |
| 52 | - | - | 0.65 | 1227.2 | 0 | 797.7 | 0 | 0 |

2. Axial + negative flexure about major axis

Section input data:

Section height h based on major axis: 16 in

Section width b based on major axis: 16 in

Rectangular compression block factors:

$$4000 \text{ psi} \leq f_c \leq 8000 \text{ psi} \rightarrow \beta_1 = 0.85 - \left(\frac{f_c - 4000}{1000} \right) \cdot 0.05 = 0.85 - \left(\frac{5000 - 4000}{1000} \right) \cdot 0.05 = 0.80$$

Section Rebar

| Depth di (in) | bar diameter (in) | bar area Asi (in^2) |
|---------------|-------------------|---------------------|
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |

Calculation of plastic centroid location

$$\bar{x}_p = \frac{f'_c \cdot A_g \cdot (h - \text{neutral axis}) + \sum F_{sci} \cdot a_{sci} + \sum F_{si} \cdot a_{si}}{f'_c \cdot A_g + f_{sy} \cdot A_s + f_{sy} \cdot \dot{A}_s} =$$

$$= \frac{5000.00 \cdot 256.00 \cdot (16 - 8) + 3240000.00 + 600000.00}{5000.00 \cdot 256.00 + 60000.00 \cdot 4 + 60000.00 \cdot 4.00} = 8.00 \text{ in}$$

M-N interaction values

| Iter. | c (in) | a (in) | φ | Pn (Kip) | Mn (Kip-ft) | φPn (Kip) | φMn (Kip-ft) | e (in) |
|-----------------|--------|--------|------|----------|-------------|-----------|--------------|--------|
| Tension Control | | | | | | | | |
| 1 | - | - | 0.9 | -480.0 | 0 | -432.0 | 0 | 0 |
| 2 | 0.32 | 0.26 | 0.90 | -462.6 | 11.4 | -416.4 | 10.3 | -0.3 |
| 3 | 0.64 | 0.51 | 0.90 | -445.2 | 22.4 | -400.7 | 20.2 | -0.6 |
| 4 | 0.96 | 0.77 | 0.90 | -427.8 | 33.1 | -385.1 | 29.8 | -0.9 |
| 5 | 1.28 | 1.02 | 0.90 | -410.5 | 43.4 | -369.4 | 39.1 | -1.3 |
| 6 | 1.60 | 1.28 | 0.90 | -349.5 | 73.3 | -314.6 | 66.0 | -2.5 |
| 7 | 1.92 | 1.53 | 0.90 | -241.4 | 124.4 | -217.3 | 112.0 | -6.2 |
| 8 | 2.24 | 1.79 | 0.90 | -159.2 | 163.3 | -143.3 | 147.0 | -12.3 |
| 9 | 2.56 | 2.05 | 0.90 | -93.2 | 194.4 | -83.9 | 175.0 | -25.0 |
| 10 | 2.88 | 2.30 | 0.90 | -38.0 | 220.2 | -34.2 | 198.2 | -69.5 |
| 11 | 3.20 | 2.56 | 0.90 | 9.6 | 242.1 | 8.7 | 217.9 | 301.5 |
| 12 | 3.52 | 2.81 | 0.90 | 51.8 | 261.2 | 46.6 | 235.0 | 60.5 |
| 13 | 3.84 | 3.07 | 0.90 | 89.8 | 277.9 | 80.8 | 250.1 | 37.2 |
| 14 | 4.15 | 3.32 | 0.90 | 124.6 | 292.9 | 112.2 | 263.6 | 28.2 |
| 15 | 4.47 | 3.58 | 0.90 | 157.0 | 306.3 | 141.3 | 275.7 | 23.4 |
| 16 | 4.79 | 3.84 | 0.90 | 187.3 | 318.5 | 168.6 | 286.7 | 20.4 |
| 17 | 5.11 | 4.09 | 0.89 | 216.0 | 329.6 | 193.0 | 294.4 | 18.3 |
| 18 | 5.43 | 4.35 | 0.85 | 243.4 | 339.6 | 207.8 | 289.9 | 16.7 |
| 19 | 5.75 | 4.60 | 0.82 | 269.7 | 348.8 | 220.7 | 285.4 | 15.5 |
| 20 | 6.07 | 4.86 | 0.79 | 295.1 | 357.2 | 232.1 | 280.9 | 14.5 |
| 21 | 6.39 | 5.11 | 0.76 | 319.6 | 364.8 | 242.3 | 276.6 | 13.7 |
| 22 | 6.71 | 5.37 | 0.73 | 343.5 | 371.8 | 251.5 | 272.3 | 13.0 |
| 23 | 7.03 | 5.62 | 0.71 | 366.8 | 378.1 | 260.0 | 268.1 | 12.4 |
| 24 | 7.35 | 5.88 | 0.69 | 389.5 | 383.9 | 267.8 | 263.9 | 11.8 |

| | | | | | | | | |
|---------------------|-------|-------|------|--------|-------|-------|-------|------|
| 25 | 7.67 | 6.14 | 0.67 | 411.8 | 389.0 | 275.1 | 259.9 | 11.3 |
| 26 | 7.99 | 6.39 | 0.65 | 433.8 | 393.6 | 281.9 | 255.8 | 10.9 |
| Compression Control | | | | | | | | |
| 27 | 8.31 | 6.65 | 0.65 | 474.7 | 385.8 | 308.6 | 250.7 | 9.8 |
| 28 | 8.63 | 6.90 | 0.65 | 513.2 | 377.9 | 333.6 | 245.6 | 8.8 |
| 29 | 8.95 | 7.16 | 0.65 | 550.1 | 370.4 | 357.6 | 240.8 | 8.1 |
| 30 | 9.27 | 7.42 | 0.65 | 585.6 | 363.1 | 380.7 | 236.0 | 7.4 |
| 31 | 9.59 | 7.67 | 0.65 | 620.0 | 356.0 | 403.0 | 231.4 | 6.9 |
| 32 | 9.91 | 7.93 | 0.65 | 653.3 | 349.1 | 424.6 | 226.9 | 6.4 |
| 33 | 10.23 | 8.19 | 0.65 | 685.5 | 342.2 | 445.6 | 222.4 | 6.0 |
| 34 | 10.55 | 8.44 | 0.65 | 716.9 | 335.3 | 466.0 | 218.0 | 5.6 |
| 35 | 10.87 | 8.70 | 0.65 | 747.5 | 328.5 | 485.8 | 213.5 | 5.3 |
| 36 | 11.19 | 8.96 | 0.65 | 777.3 | 321.6 | 505.2 | 209.0 | 5.0 |
| 37 | 11.51 | 9.21 | 0.65 | 806.4 | 314.7 | 524.1 | 204.5 | 4.7 |
| 38 | 11.83 | 9.47 | 0.65 | 834.8 | 307.7 | 542.6 | 200.0 | 4.4 |
| 39 | 12.16 | 9.72 | 0.65 | 862.7 | 300.6 | 560.8 | 195.4 | 4.2 |
| 40 | 12.48 | 9.98 | 0.65 | 890.1 | 293.3 | 578.6 | 190.7 | 4.0 |
| 41 | 12.80 | 10.24 | 0.65 | 916.9 | 285.9 | 596.0 | 185.9 | 3.7 |
| 42 | 13.12 | 10.49 | 0.65 | 943.3 | 278.4 | 613.2 | 181.0 | 3.5 |
| 43 | 13.44 | 10.75 | 0.65 | 969.3 | 270.7 | 630.1 | 175.9 | 3.4 |
| 44 | 13.76 | 11.01 | 0.65 | 994.9 | 262.8 | 646.7 | 170.8 | 3.2 |
| 45 | 14.08 | 11.26 | 0.65 | 1020.1 | 254.6 | 663.1 | 165.5 | 3.0 |
| 46 | 14.40 | 11.52 | 0.65 | 1045.0 | 246.3 | 679.2 | 160.1 | 2.8 |
| 47 | 14.72 | 11.77 | 0.65 | 1069.5 | 237.8 | 695.2 | 154.5 | 2.7 |
| 48 | 15.04 | 12.03 | 0.65 | 1093.7 | 229.0 | 710.9 | 148.8 | 2.5 |
| 49 | 15.36 | 12.29 | 0.65 | 1117.7 | 219.9 | 726.5 | 143.0 | 2.4 |
| 50 | 15.68 | 12.54 | 0.65 | 1141.3 | 210.7 | 741.9 | 136.9 | 2.2 |
| 51 | 16.00 | 12.80 | 0.65 | 1164.8 | 201.1 | 757.1 | 130.7 | 2.1 |
| 52 | - | - | 0.65 | 1227.2 | 0 | 797.7 | 0 | 0 |

3. Axial + positive flexure about minor axis

Section input data:

Section height h based on minor axis: 16 in

Section width b based on minor axis: 16 in

Rectangular compression block factors:

$$4000 \text{ psi} \leq f_c \leq 8000 \text{ psi} \rightarrow \beta_1 = 0.85 - \left(\frac{f_c - 4000}{1000} \right) \cdot 0.05 = 0.85 - \left(\frac{5000 - 4000}{1000} \right) \cdot 0.05 = 0.80$$

Section Rebar

| Depth di (in) | bar diameter (in) | bar area Asi (in^2) |
|---------------|-------------------|---------------------|
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 9.83 | 1.13 | 1.00 |
| 9.83 | 1.13 | 1.00 |
| 6.17 | 1.13 | 1.00 |
| 6.17 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |

Calculation of plastic centroid location

$$\bar{x}_p = \frac{f'_c \cdot A_g \cdot (h - \text{neutral axis}) + \sum F_{sci} \cdot a_{sci} + \sum F_{si} \cdot a_{si}}{f'_c \cdot A_g + f_{sy} \cdot A_s + f_{sy} \cdot \dot{A}_s} =$$

$$= \frac{5000.00 \cdot 256.00 \cdot (16 - 8) + 2800000.00 + 1040000.00}{5000.00 \cdot 256.00 + 60000.00 \cdot 4 + 60000.00 \cdot 4.00} = 8.00 \text{ in}$$

M-N interaction values

| Iter. | c (in) | a (in) | φ | Pn (Kip) | Mn (Kip-ft) | φPn (Kip) | φMn (Kip-ft) | e (in) |
|-----------------|--------|--------|------|----------|-------------|-----------|--------------|--------|
| Tension Control | | | | | | | | |
| 1 | - | - | 0.9 | -480.0 | 0 | -432.0 | 0 | 0 |
| 2 | 0.28 | 0.22 | 0.90 | -465.0 | 9.9 | -418.5 | 8.9 | -0.3 |
| 3 | 0.55 | 0.44 | 0.90 | -450.0 | 19.5 | -405.0 | 17.5 | -0.5 |
| 4 | 0.83 | 0.66 | 0.90 | -434.9 | 28.8 | -391.4 | 25.9 | -0.8 |
| 5 | 1.10 | 0.88 | 0.90 | -419.9 | 37.9 | -377.9 | 34.1 | -1.1 |
| 6 | 1.38 | 1.10 | 0.90 | -404.9 | 46.6 | -364.4 | 42.0 | -1.4 |
| 7 | 1.66 | 1.33 | 0.90 | -358.4 | 69.6 | -322.5 | 62.6 | -2.3 |
| 8 | 1.93 | 1.55 | 0.90 | -305.8 | 95.0 | -275.2 | 85.5 | -3.7 |
| 9 | 2.21 | 1.77 | 0.90 | -262.7 | 115.8 | -236.4 | 104.2 | -5.3 |
| 10 | 2.49 | 1.99 | 0.90 | -225.8 | 133.5 | -203.2 | 120.1 | -7.1 |
| 11 | 2.76 | 2.21 | 0.90 | -193.3 | 148.9 | -173.9 | 134.0 | -9.2 |
| 12 | 3.04 | 2.43 | 0.90 | -163.9 | 162.6 | -147.5 | 146.3 | -11.9 |
| 13 | 3.31 | 2.65 | 0.90 | -137.0 | 174.9 | -123.3 | 157.4 | -15.3 |
| 14 | 3.59 | 2.87 | 0.90 | -111.8 | 186.1 | -100.6 | 167.5 | -20.0 |
| 15 | 3.87 | 3.09 | 0.90 | -71.7 | 198.8 | -64.5 | 178.9 | -33.3 |
| 16 | 4.14 | 3.31 | 0.90 | -30.6 | 211.1 | -27.6 | 190.0 | -82.7 |
| 17 | 4.42 | 3.54 | 0.90 | 7.1 | 222.3 | 6.4 | 200.1 | 373.3 |
| 18 | 4.70 | 3.76 | 0.90 | 42.2 | 232.6 | 38.0 | 209.3 | 66.1 |
| 19 | 4.97 | 3.98 | 0.90 | 75.1 | 242.1 | 67.6 | 217.9 | 38.7 |
| 20 | 5.25 | 4.20 | 0.88 | 106.1 | 250.8 | 92.9 | 219.7 | 28.4 |
| 21 | 5.52 | 4.42 | 0.84 | 135.5 | 258.9 | 114.2 | 218.3 | 22.9 |
| 22 | 5.80 | 4.64 | 0.81 | 163.5 | 266.4 | 133.0 | 216.6 | 19.5 |
| 23 | 6.08 | 4.86 | 0.79 | 202.8 | 271.4 | 159.4 | 213.4 | 16.1 |
| 24 | 6.35 | 5.08 | 0.76 | 240.8 | 275.9 | 183.4 | 210.1 | 13.7 |

| | | | | | | | | |
|---------------------|-------|-------|------|--------|-------|-------|-------|------|
| 25 | 6.63 | 5.30 | 0.74 | 277.0 | 280.1 | 204.6 | 206.9 | 12.1 |
| 26 | 6.90 | 5.52 | 0.72 | 311.4 | 284.0 | 223.6 | 203.9 | 10.9 |
| Compression Control | | | | | | | | |
| 27 | 7.27 | 5.81 | 0.69 | 354.5 | 288.5 | 245.7 | 199.9 | 9.8 |
| 28 | 7.63 | 6.11 | 0.67 | 395.4 | 292.6 | 265.0 | 196.1 | 8.9 |
| 29 | 8.00 | 6.40 | 0.65 | 434.7 | 296.0 | 282.5 | 192.4 | 8.2 |
| 30 | 8.36 | 6.69 | 0.65 | 482.8 | 292.1 | 313.8 | 189.9 | 7.3 |
| 31 | 8.72 | 6.98 | 0.65 | 528.2 | 288.2 | 343.3 | 187.3 | 6.5 |
| 32 | 9.09 | 7.27 | 0.65 | 571.5 | 284.3 | 371.5 | 184.8 | 6.0 |
| 33 | 9.45 | 7.56 | 0.65 | 613.1 | 280.3 | 398.5 | 182.2 | 5.5 |
| 34 | 9.82 | 7.85 | 0.65 | 653.0 | 276.1 | 424.4 | 179.5 | 5.1 |
| 35 | 10.18 | 8.14 | 0.65 | 691.5 | 271.9 | 449.5 | 176.7 | 4.7 |
| 36 | 10.54 | 8.43 | 0.65 | 728.7 | 267.4 | 473.6 | 173.8 | 4.4 |
| 37 | 10.91 | 8.73 | 0.65 | 764.7 | 262.7 | 497.1 | 170.8 | 4.1 |
| 38 | 11.27 | 9.02 | 0.65 | 799.7 | 257.8 | 519.8 | 167.6 | 3.9 |
| 39 | 11.63 | 9.31 | 0.65 | 833.7 | 252.7 | 541.9 | 164.2 | 3.6 |
| 40 | 12.00 | 9.60 | 0.65 | 866.9 | 247.2 | 563.5 | 160.7 | 3.4 |
| 41 | 12.36 | 9.89 | 0.65 | 899.3 | 241.4 | 584.5 | 156.9 | 3.2 |
| 42 | 12.73 | 10.18 | 0.65 | 930.9 | 235.4 | 605.1 | 153.0 | 3.0 |
| 43 | 13.09 | 10.47 | 0.65 | 961.9 | 229.0 | 625.3 | 148.8 | 2.9 |
| 44 | 13.45 | 10.76 | 0.65 | 992.3 | 222.2 | 645.0 | 144.4 | 2.7 |
| 45 | 13.82 | 11.05 | 0.65 | 1022.2 | 215.1 | 664.4 | 139.8 | 2.5 |
| 46 | 14.18 | 11.34 | 0.65 | 1051.5 | 207.7 | 683.5 | 135.0 | 2.4 |
| 47 | 14.54 | 11.64 | 0.65 | 1080.3 | 199.9 | 702.2 | 129.9 | 2.2 |
| 48 | 14.91 | 11.93 | 0.65 | 1108.7 | 191.6 | 720.7 | 124.6 | 2.1 |
| 49 | 15.27 | 12.22 | 0.65 | 1136.7 | 183.1 | 738.9 | 119.0 | 1.9 |
| 50 | 15.64 | 12.51 | 0.65 | 1164.3 | 174.1 | 756.8 | 113.1 | 1.8 |
| 51 | 16.00 | 12.80 | 0.65 | 1191.6 | 164.7 | 774.5 | 107.0 | 1.7 |
| 52 | - | - | 0.65 | 1227.2 | 0 | 797.7 | 0 | 0 |

4. Axial + negative flexure about minor axis

Section input data:

Section height h based on minor axis: 16 in

Section width b based on minor axis: 16 in

Rectangular compression block factors:

$$4000 \text{ psi} \leq f_c \leq 8000 \text{ psi} \rightarrow \beta_1 = 0.85 - \left(\frac{f_c - 4000}{1000} \right) \cdot 0.05 = 0.85 - \left(\frac{5000 - 4000}{1000} \right) \cdot 0.05 = 0.80$$

Section Rebar

| Depth di (in) | bar diameter (in) | bar area Asi (in^2) |
|---------------|-------------------|---------------------|
| 13.50 | 1.13 | 1.00 |
| 13.50 | 1.13 | 1.00 |
| 9.83 | 1.13 | 1.00 |
| 9.83 | 1.13 | 1.00 |
| 6.17 | 1.13 | 1.00 |
| 6.17 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |
| 2.50 | 1.13 | 1.00 |

Calculation of plastic centroid location

$$\bar{x}_p = \frac{f'_c \cdot A_g \cdot (h - \text{neutral axis}) + \sum F_{sci} \cdot a_{sci} + \sum F_{si} \cdot a_{si}}{f'_c \cdot A_g + f_{sy} \cdot A_s + f_{sy} \cdot \dot{A}_s} =$$

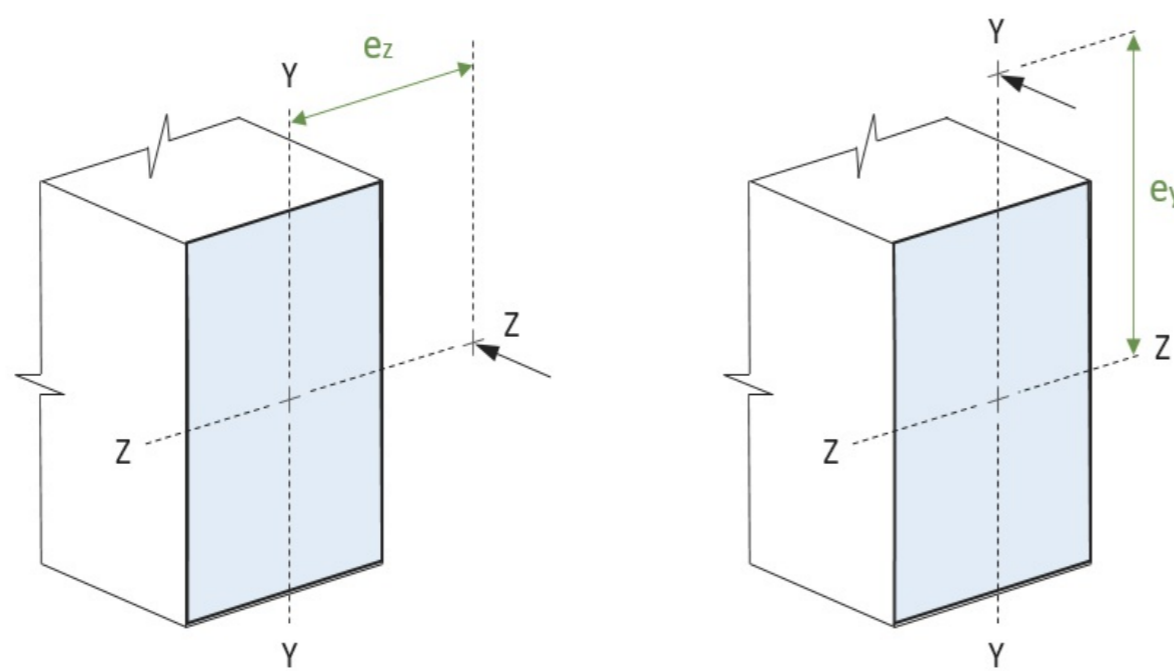
$$= \frac{5000.00 \cdot 256.00 \cdot (16 - 8) + 2800000.00 + 1040000.00}{5000.00 \cdot 256.00 + 60000.00 \cdot 4 + 60000.00 \cdot 4.00} = 8.00 \text{ in}$$

M-N interaction values

| Iter. | c (in) | a (in) | φ | Pn (Kip) | Mn (Kip-ft) | φPn (Kip) | φMn (Kip-ft) | e (in) |
|-----------------|--------|--------|------|----------|-------------|-----------|--------------|--------|
| Tension Control | | | | | | | | |
| 1 | - | - | 0.9 | -480.0 | 0 | -432.0 | 0 | 0 |
| 2 | 0.28 | 0.22 | 0.90 | -465.0 | 9.9 | -418.5 | 8.9 | -0.3 |
| 3 | 0.55 | 0.44 | 0.90 | -450.0 | 19.5 | -405.0 | 17.5 | -0.5 |
| 4 | 0.83 | 0.66 | 0.90 | -434.9 | 28.8 | -391.4 | 25.9 | -0.8 |
| 5 | 1.10 | 0.88 | 0.90 | -419.9 | 37.9 | -377.9 | 34.1 | -1.1 |
| 6 | 1.38 | 1.10 | 0.90 | -404.9 | 46.6 | -364.4 | 42.0 | -1.4 |
| 7 | 1.66 | 1.33 | 0.90 | -358.4 | 69.6 | -322.5 | 62.6 | -2.3 |
| 8 | 1.93 | 1.55 | 0.90 | -305.8 | 95.0 | -275.2 | 85.5 | -3.7 |
| 9 | 2.21 | 1.77 | 0.90 | -262.7 | 115.8 | -236.4 | 104.2 | -5.3 |
| 10 | 2.49 | 1.99 | 0.90 | -225.8 | 133.5 | -203.2 | 120.1 | -7.1 |
| 11 | 2.76 | 2.21 | 0.90 | -193.3 | 148.9 | -173.9 | 134.0 | -9.2 |
| 12 | 3.04 | 2.43 | 0.90 | -163.9 | 162.6 | -147.5 | 146.3 | -11.9 |
| 13 | 3.31 | 2.65 | 0.90 | -137.0 | 174.9 | -123.3 | 157.4 | -15.3 |
| 14 | 3.59 | 2.87 | 0.90 | -111.8 | 186.1 | -100.6 | 167.5 | -20.0 |
| 15 | 3.87 | 3.09 | 0.90 | -71.7 | 198.8 | -64.5 | 178.9 | -33.3 |
| 16 | 4.14 | 3.31 | 0.90 | -30.6 | 211.1 | -27.6 | 190.0 | -82.7 |
| 17 | 4.42 | 3.54 | 0.90 | 7.1 | 222.3 | 6.4 | 200.1 | 373.3 |
| 18 | 4.70 | 3.76 | 0.90 | 42.2 | 232.6 | 38.0 | 209.3 | 66.1 |
| 19 | 4.97 | 3.98 | 0.90 | 75.1 | 242.1 | 67.6 | 217.9 | 38.7 |
| 20 | 5.25 | 4.20 | 0.88 | 106.1 | 250.8 | 92.9 | 219.7 | 28.4 |
| 21 | 5.52 | 4.42 | 0.84 | 135.5 | 258.9 | 114.2 | 218.3 | 22.9 |
| 22 | 5.80 | 4.64 | 0.81 | 163.5 | 266.4 | 133.0 | 216.6 | 19.5 |
| 23 | 6.08 | 4.86 | 0.79 | 202.8 | 271.4 | 159.4 | 213.4 | 16.1 |
| 24 | 6.35 | 5.08 | 0.76 | 240.8 | 275.9 | 183.4 | 210.1 | 13.7 |

| | | | | | | | | |
|---------------------|-------|-------|------|--------|-------|-------|-------|------|
| 25 | 6.63 | 5.30 | 0.74 | 277.0 | 280.1 | 204.6 | 206.9 | 12.1 |
| 26 | 6.90 | 5.52 | 0.72 | 311.4 | 284.0 | 223.6 | 203.9 | 10.9 |
| Compression Control | | | | | | | | |
| 27 | 7.27 | 5.81 | 0.69 | 354.5 | 288.5 | 245.7 | 199.9 | 9.8 |
| 28 | 7.63 | 6.11 | 0.67 | 395.4 | 292.6 | 265.0 | 196.1 | 8.9 |
| 29 | 8.00 | 6.40 | 0.65 | 434.7 | 296.0 | 282.5 | 192.4 | 8.2 |
| 30 | 8.36 | 6.69 | 0.65 | 482.8 | 292.1 | 313.8 | 189.9 | 7.3 |
| 31 | 8.72 | 6.98 | 0.65 | 528.2 | 288.2 | 343.3 | 187.3 | 6.5 |
| 32 | 9.09 | 7.27 | 0.65 | 571.5 | 284.3 | 371.5 | 184.8 | 6.0 |
| 33 | 9.45 | 7.56 | 0.65 | 613.1 | 280.3 | 398.5 | 182.2 | 5.5 |
| 34 | 9.82 | 7.85 | 0.65 | 653.0 | 276.1 | 424.4 | 179.5 | 5.1 |
| 35 | 10.18 | 8.14 | 0.65 | 691.5 | 271.9 | 449.5 | 176.7 | 4.7 |
| 36 | 10.54 | 8.43 | 0.65 | 728.7 | 267.4 | 473.6 | 173.8 | 4.4 |
| 37 | 10.91 | 8.73 | 0.65 | 764.7 | 262.7 | 497.1 | 170.8 | 4.1 |
| 38 | 11.27 | 9.02 | 0.65 | 799.7 | 257.8 | 519.8 | 167.6 | 3.9 |
| 39 | 11.63 | 9.31 | 0.65 | 833.7 | 252.7 | 541.9 | 164.2 | 3.6 |
| 40 | 12.00 | 9.60 | 0.65 | 866.9 | 247.2 | 563.5 | 160.7 | 3.4 |
| 41 | 12.36 | 9.89 | 0.65 | 899.3 | 241.4 | 584.5 | 156.9 | 3.2 |
| 42 | 12.73 | 10.18 | 0.65 | 930.9 | 235.4 | 605.1 | 153.0 | 3.0 |
| 43 | 13.09 | 10.47 | 0.65 | 961.9 | 229.0 | 625.3 | 148.8 | 2.9 |
| 44 | 13.45 | 10.76 | 0.65 | 992.3 | 222.2 | 645.0 | 144.4 | 2.7 |
| 45 | 13.82 | 11.05 | 0.65 | 1022.2 | 215.1 | 664.4 | 139.8 | 2.5 |
| 46 | 14.18 | 11.34 | 0.65 | 1051.5 | 207.7 | 683.5 | 135.0 | 2.4 |
| 47 | 14.54 | 11.64 | 0.65 | 1080.3 | 199.9 | 702.2 | 129.9 | 2.2 |
| 48 | 14.91 | 11.93 | 0.65 | 1108.7 | 191.6 | 720.7 | 124.6 | 2.1 |
| 49 | 15.27 | 12.22 | 0.65 | 1136.7 | 183.1 | 738.9 | 119.0 | 1.9 |
| 50 | 15.64 | 12.51 | 0.65 | 1164.3 | 174.1 | 756.8 | 113.1 | 1.8 |
| 51 | 16.00 | 12.80 | 0.65 | 1191.6 | 164.7 | 774.5 | 107.0 | 1.7 |
| 52 | - | - | 0.65 | 1227.2 | 0 | 797.7 | 0 | 0 |

5. Strength check



Section input data:

Actual axial force $P = 100.00$ Kip
 Actual bending moment about major axis $M_{uz} = 10.00$ Kip-ft
 Actual bending moment about minor axis $M_{uy} = 10.00$ Kip-ft
 Eccentricity of actual forces along major axis $e_y = 1.2000$ in
 Eccentricity of actual forces along minor axis $e_z = 1.2000$ in
 Limited axial force (Mx-N axis plane) $\phi \cdot P_{nz} = 774.18$ Kip
 Limited axial force (My-N axis plane) $\phi \cdot P_{ny} = 780.93$ Kip
 Limited bending moment about major axis $\phi \cdot M_{nz} = 75.71$ Kip-ft
 Limited bending moment about minor axis $\phi \cdot M_{ny} = 77.45$ Kip-ft

Axial + biaxial bending check case for rectangular section

$$\alpha_n = 0.7 + \frac{1.7 \cdot P_u}{0.6 \cdot P_{no}} = 0.7 + \frac{1.7 \cdot 100}{0.6 \cdot 797.68} = 1.06$$

$$1.0 \geq \alpha_n \leq 2.0$$

$$\left[\frac{M_z}{\phi \cdot M_{ux}} \right]^{\alpha_n} + \left[\frac{M_y}{\phi \cdot M_{uy}} \right]^{\alpha_n} = \left[\frac{10.00}{75.71} \right]^{1.06} + \left[\frac{10.00}{77.45} \right]^{1.06} = 0.23 \leq 1.0$$

Axial compression check case

$$P_n = 100.00 \text{ Kip} \leq \phi \cdot P_{no} = 797.68 \text{ Kip}$$

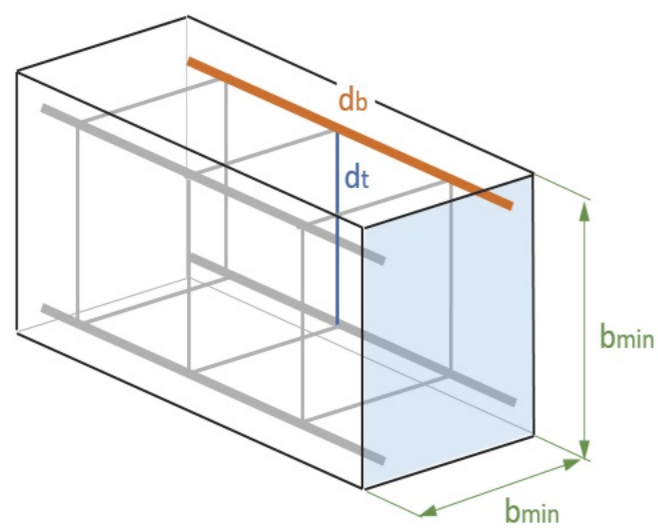
STATUS OK!

STATUS OK!

CHAPTER 9 (Section 9.7)

Lateral support check for compression steel

BUCKLING OF SLENDER REINFORCING BARS



Section input data:

Spacing of the stirrups $s = 10$ in

Diameter of the main bars $16 \cdot d_b = 16 \cdot 1.128 = 18.048$ in

Diameter of the transverse reinforcement (stirrups) $48 \cdot d_t = 48 \cdot 0.5 = 24$ in

Smaller dimension of the beam section $b_{min} = 16$ in

Calculate maximum spacing of the stirrups for purpose of buckling bar stability

$$s_{max} = \min[16 \cdot d_b, 48 \cdot d_t, b_{min}] = 16 \text{ in}$$

$$s = 10 \text{ in} \leq s_{max} = 16 \text{ in}$$

STATUS OK!